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Heat transfer in the flow of a cold, two-dimensional draining sheet over a hot, horizontal cylinder

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1. Introduction

The draining flow under gravity falling across a horizontal cylinder occurs frequently in a variety of industrial heat-transfer applications, such as heat exchange in water desalinization apparatus [1], in absorption coolers [2], in power and process condensers [3] and in thin film cooling [4]. In order to understand the operations and, in particular, the efficiency of these processes, it is important to have a detailed study of such a flow and has been the subject of numerous theoretical and experimental research studies reported in the literature. Comprehensive literature reviews of such draining flow studies have been made by Thome [5] and Ribatski and Jacobi [6] to reveal fully the developments in this subject.

Mitrovic [7] pointed out that the transition of draining flow patterns, such as droplet, column or sheet, was dependent on flow rates. At a high flow rate, the draining flow pattern may be in the form of sheet. It is natural to a closer inspection of the draining sheet flow pattern. However previous studies of such draining sheet flow over a horizontal cylinder have been concerned with the purely hydrodynamic problem. Abdelghaffer et al. [8] obtained an approximation using the Pohlhausen integral momentum technique, which assumed an approximate velocity profile across the thickness of the film; Hunt [9] obtained a numerical solution using the modified Keller box method, which accommodated the outer, free boundary. Heat-transfer characteristics of the flow have not been considered.

In this paper, the heat transfer in the flow of a cold, twodimensional draining sheet over a hot, horizontal cylinder is inves-

ABSTRACT

The paper considers heat transfer characteristics of thin film flow over a hot horizontal cylinder resulting from a cold vertical sheet of liquid falling onto the surface. The underlying physical features of the developing film thickness, velocity and temperature distributions have been illustrated by numerical solutions of high accuracy for large Reynolds numbers using the modified Keller box method. The solutions for film thickness distribution are good agreement with those obtained using the Pohlhausen integral momentum technique thus providing a basic confirmation of the validity of the results presented. © 2008 Elsevier Masson SAS. All rights reserved.

tigated. The accurate and comprehensive numerical solutions for both velocity and temperature distributions is obtained.

2. Modelling

The problem to be examined concerns the film cooling which occurs when a cold vertically draining sheet strikes a hot horizontal cylinder. Although a sheet of fluid draining under gravity is accelerated and thin at impact [10,11], it is reasonable to model the associated volume flow as a sheet of uniform velocity U_0 , uniform temperature T_0 and semi-thickness H_0 as is illustrated in Fig. 1. The notation $Q = U_0H_0$ is introduced for the flow rate and a film Reynolds number may be defined as $R_e = U_0a/\nu$ based on the cylinder radius where ν is the kinematic viscosity of the fluid and *a* is the radius of the cylinder.

3. Governing equations

The flow under investigation has been modelled as a steady, two-dimensional flow of incompressible fluid. In the absence of viscous dissipation, the equations expressing conservation of mass, momentum and energy are consequently

$$\begin{aligned} \frac{\partial v_{\theta}}{\partial \theta} &+ \frac{\partial}{\partial r} (r v_{r}) = 0, \end{aligned} \tag{1} \\ \frac{v_{\theta}}{r} \frac{\partial v_{\theta}}{\partial \theta} &+ v_{r} \frac{\partial v_{\theta}}{\partial r} + \frac{v_{r} v_{\theta}}{r} \\ &= g \sin \theta - \frac{1}{\rho r} \frac{dP}{d\theta} \\ &+ \frac{\mu}{\rho} \bigg[\frac{1}{r} \frac{\partial}{\partial r} \bigg(r \frac{\partial v_{\theta}}{\partial r} \bigg) + \frac{1}{r^{2}} \frac{\partial^{2} v_{\theta}}{\partial \theta^{2}} + \frac{2}{r^{2}} \frac{\partial v_{r}}{\partial \theta} - \frac{v_{\theta}}{r^{2}} \bigg], \end{aligned} \tag{2}$$

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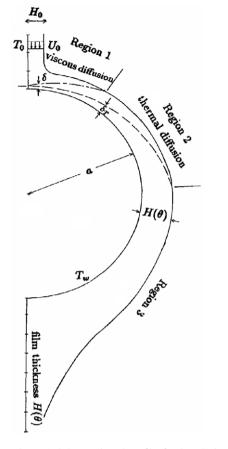


Fig. 1. The vertical sheet and resultant film for the cylinder.

$$\frac{\nu_{\theta}}{r}\frac{\partial T}{\partial \theta} + \nu_{r}\frac{\partial T}{\partial r} = \frac{k}{\rho C_{p}} \left[\frac{1}{r}\frac{\partial}{\partial r} \left(r\frac{\partial T}{\partial r} \right) + \frac{1}{r^{2}}\frac{\partial^{2} T}{\partial \theta^{2}} \right]$$
(3)

where $\vec{V} = (v_{\theta}, v_r)$ are velocity components associated with cylindrical coordinates (θ, r) measured the angular displacement from the top of the cylinder and the radial distance from the centre of the cylinder respectively. ρ , μ , C_p and k are the density, dynamic viscosity, specific heat at constant pressure and the thermal conductivity of the cooling fluid in the sheet respectively. T and P are respectively the temperature and pressure within the fluid.

In the specified physical setting, the equations are to be solved subject to the following conditions.

(i) The no slip boundary condition at the cylinder wall requires that

$$\nu_{\theta} = \nu_r = 0 \quad \text{on } r = a, \ 0 \leqslant \theta \leqslant \pi.$$
(4)

(ii) The temperature at the cylinder wall is assumed constant as T_w , say i.e.

$$T = T_w \quad \text{on } r = a, \ 0 \leqslant \theta \leqslant \pi.$$
⁽⁵⁾

(iii) On the free surface of the film, prescribed by $r = a + H(\theta)$, the shearing stress may be assumed negligible and consequently

$$r\frac{\partial}{\partial r}\left(\frac{\nu_{\theta}}{r}\right) + \frac{1}{r}\frac{\partial\nu_{r}}{\partial\theta} = 0 \quad \text{at } r = a + H(\theta), \ 0 \leqslant \theta \leqslant \pi.$$
(6)

(iv) Similarly, in a film cooling environment such as water surrounded by air, it may be assumed that there is negligible heat flux on the free surface and hence that

$$\frac{\partial T}{\partial r} = 0$$
 at $r = a + H(\theta), \ 0 \le \theta \le \pi$. (7)

(v) Once an overall flow rate $Q = U_0 H_0$ has been prescribed, a conservation of volume flow constraint at any given θ station leads to the condition

 $a{+}H(\theta)$

$$\int_{a} v_{\theta}(\theta, r) \, dr = \text{constant} = H_0 U_0 \quad \text{for } 0 \leqslant \theta \leqslant \pi \,. \tag{8}$$

Under the assumption that the film thickness remains thin relative to a characteristic horizontal dimension, a boundary layer treatment of the equations leads to significant simplification. The following non-dimensional variables are introduced

$$x = \theta, \quad y = \frac{R_e^{1/2}(r-a)}{a}, \quad \bar{h}(x) = \frac{R_e^{1/2}H(\theta)}{a},$$
$$\bar{U} = \frac{\nu_{\theta}}{U_0}, \quad \bar{V} = \frac{R_e^{1/2}\nu_r}{U_0}, \quad \bar{\phi} = \frac{T-T_w}{T_0 - T_w}, \quad p = \frac{P}{\rho U_0^2}.$$
(9)

In the limit $R_e \to +\infty$ with *x* remaining O(1) and after neglecting terms of O($\frac{1}{R_e^{1/2}}$) compared with unity, the following equations are obtained

$$\frac{\partial \bar{U}}{\partial x} + \frac{\partial \bar{V}}{\partial y} = 0, \tag{10}$$

$$\bar{U}\frac{\partial\bar{U}}{\partial x} + \bar{V}\frac{\partial\bar{U}}{\partial y} = \frac{1}{F_r}\sin x - \frac{dp}{dx} + \frac{\partial^2\bar{U}}{\partial y^2},\tag{11}$$

$$P_r\left(\bar{U}\frac{\partial\bar{\phi}}{\partial x} + \bar{V}\frac{\partial\bar{\phi}}{\partial y}\right) = \frac{\partial^2\bar{\phi}}{\partial y^2}$$
(12)

where $P_r = \nu/\kappa$ is the Prandtl number with ν the kinematic viscosity μ/ρ and κ the thermometric conductivity $k/(\rho C_p)$, and $F_r = U_0^2/(ag)$ is the Froude number based on the sheet velocity on its surface. In common with standard boundary layer theory equation implies that the pressure across the film remains constant. In the absence of external pressure gradients and with zero shear assumed on the free surface, the pressure term in (11) is identically zero.

The boundary conditions now read

(i)
$$\overline{U} = \overline{V} = 0$$
 on $y = 0, \ 0 \leq x \leq \pi$, (13)

(ii)
$$\bar{\phi} = 0$$
 on $y = 0, \ 0 \leq x \leq \pi$, (14)

(iii)
$$\frac{\partial U}{\partial y} = 0$$
 at $y = \bar{h}(x), \ 0 \le x \le \pi$, (15)

(iv)
$$\frac{\partial \phi}{\partial y} = 0$$
 at $y = \bar{h}(x), \ 0 \le x \le \pi$, (16)

(v)
$$\int_{0}^{h(x)} \bar{U} \, dy = \frac{R_e^{1/2} H_0}{a} \quad \text{for } 0 \le x \le \pi \,.$$
 (17)

Approximations of equations equivalent to (10)-(11) under boundary conditions (13), (15) and (17) have been outlined by Abdelghaffer et al. [8].

4. Numerical solutions

The continuity equation (10) can be eliminated by introducing a stream function ψ defined by

$$\bar{U} = \frac{\partial \psi}{\partial y}, \qquad \bar{V} = -\frac{\partial \psi}{\partial x}.$$
 (18)

Substituting Eq. (18) into (10)–(17) gives

$$\frac{\partial^3 \psi}{\partial y^3} + \frac{1}{F_r} \sin x = \frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2},$$
(19)

$$\frac{\partial^2 \phi}{\partial y^2} = P_r \left(\frac{\partial \psi}{\partial y} \frac{\partial \phi}{\partial x} - \frac{\partial \psi}{\partial x} \frac{\partial \phi}{\partial y} \right)$$
(20)

subject to boundary conditions

$$\psi = 0, \quad \frac{\partial \psi}{\partial y} = 0, \quad \bar{\phi} = 0 \quad \text{on } y = 0, \quad 0 \leqslant x \leqslant \pi,$$
 (21)

$$\psi = \frac{R_e^{1/2} H_0}{a}, \quad \frac{\partial^2 \psi}{\partial y^2} = 0, \quad \frac{\partial \bar{\phi}}{\partial y} = 0 \quad \text{at } y = \bar{h}(x), \ 0 \le x \le \pi, \quad (22)$$

$$\bar{h} = \frac{R_e^{1/2}H_0}{a}, \quad \psi = y, \quad \bar{\phi} = 1 \quad \text{on } x = 0, \ 0 < y \le \frac{R_e^{1/2}H_0}{a}$$
 (23)

where the initial condition (23) appears due to the original initial condition

$$H = H_0, \quad v_0 = U_0, \quad T = T_0 \quad \text{on } \theta = 0, \ a < r \le a + H_0.$$
 (24)

In anticipation of the use of a Keller box method and its attractive extrapolation features the differential system (19)-(23) is re-cast as the following first order system

$$\frac{\partial \psi}{\partial y} = \bar{u},$$

$$\frac{\partial \bar{u}}{\partial y} = \bar{v},$$

$$\frac{\partial \bar{v}}{\partial y} = -\frac{1}{F_r} \sin x + \bar{u} \frac{\partial \bar{u}}{\partial x} - \bar{v} \frac{\partial \psi}{\partial x},$$

$$\frac{\partial \bar{\phi}}{\partial y} = \bar{w},$$

$$\frac{\partial \bar{w}}{\partial y} = P_r \left(\bar{u} \frac{\partial \bar{\phi}}{\partial x} - \bar{w} \frac{\partial \psi}{\partial x} \right)$$
(25)

whose boundary conditions are

$$\psi = 0, \quad \bar{u} = 0, \quad \bar{\phi} = 0 \quad \text{on } y = 0, \quad 0 \le x \le \pi,$$

$$\psi = \frac{R_e^{1/2} H_0}{a}, \quad \bar{\nu} = 0, \quad \bar{w} = 0 \quad \text{on } y = \bar{h}(x), \quad 0 \le x \le \pi,$$

$$\bar{h} = \frac{R_e^{1/2} H_0}{a}, \quad \psi = y, \quad \bar{\phi} = 1 \quad \text{on } x = 0, \quad 0 < y \le \frac{R_e^{1/2} H_0}{a}.$$
 (26)

If these equations were used as the basis of solution there would be strong comparisons between the associated algorithm and that developed in [12]. However the expectation of an initial Blasius boundary layer within the film can be assimilated into the solution scheme by further transformations. The underlying methodology of solution nevertheless remains the same. In each case the discretization is aimed at generating a simultaneous system of non-linear equations which can be solved by Newton iteration.

According to the non-dimensional transformation, the boundary layer thickness grows like $x^{1/2}$ for small *x* in the *y* direction.

The following coordinate transformation, what simultaneously maps the film thickness onto the unit interval and removes the Blasius singularity at the origin, is introduced

$$x = \xi^2, \qquad y = \frac{\xi \eta \bar{h}}{\xi + 1 - \eta}.$$

The dependent variables are transformed as

$$\psi = \frac{\xi}{\xi + 1 - \eta} f, \quad \bar{u} = u, \quad \bar{\nu} = \frac{\xi + 1 - \eta}{\xi} \nu,$$
$$\bar{\phi} = \phi, \quad \bar{w} = \frac{\xi + 1 - \eta}{\xi} w, \quad \bar{h} = h.$$

The equations to be solved now read

$$f_{\eta} = \frac{(1+\xi)hu}{\xi+1-\eta} - \frac{f}{\xi+1-\eta},$$

$$u_{\eta} = \frac{(1+\xi)hv}{\xi+1-\eta},$$

$$v_{\eta} = \frac{v}{\xi+1-\eta} - \frac{\xi^{2}(1+\xi)h\sin(\xi^{2})}{(\xi+1-\eta)^{3}F_{r}} - \frac{(1-\eta)(1+\xi)hfv}{2(\xi+1-\eta)^{4}} + \frac{\xi(1+\xi)h}{2(\xi+1-\eta)^{3}}(uu_{\xi} - vf_{\xi}),$$

$$\phi_{\eta} = \frac{(1+\xi)hw}{\xi+1-\eta},$$

$$w_{\eta} = \frac{w}{\xi+1-\eta} - \frac{P_{r}(1-\eta)(1+\xi)hfw}{2(\xi+1-\eta)^{4}} + \frac{P_{r}\xi(1+\xi)h}{2(\xi+1-\eta)^{3}}(u\phi_{\xi} - wf_{\xi})$$
(27)

subject to

$$f = 0, \quad u = 0, \quad \phi = 0 \quad \text{on } \eta = 0, \quad 0 \le \xi \le \sqrt{\pi},$$

$$f = \frac{R_e^{1/2} H_0}{a}, \quad v = 0, \quad w = 0 \quad \text{on } \eta = 1, \quad 0 \le \xi \le \sqrt{\pi},$$

$$h = \frac{R_e^{1/2} H_0}{a}, \quad f = f_0(\eta), \quad \phi = \phi_0(\eta) \quad \text{on } \xi = 0, \quad 0 < \eta \le 1 \quad (28)$$
where the initial profiles $f_0(\eta)$ and $f_0(\eta)$ are found by proting

where the initial profiles $f_0(\eta)$ and $\phi_0(\eta)$ are found by putting $\xi = 0$ and $h = \frac{R_e^{1/2}H_0}{a}$ into (27) and solving, subject to conditions $f = u = \phi = 0$ at $\eta = 0$ and u = 1, $\phi = 1$ at $\eta = 1$.

The parabolic system of equations and boundary conditions (27)–(28) has been solved by marching in the ξ -direction using a modification of the Keller box method. A non-uniform grid is placed on the domain $\xi \ge 0$, $0 \le \eta \le 1$ and the resulting difference equations are solved by Newton iteration. Solutions are obtained on different sized grids and Richardson's extrapolation used to produce results of high accuracy.

It is worth to mention to this end that the general equations describing parabolic free boundary problems arising from thin film flows [12] are

$$\frac{\partial^3 \psi}{\partial y^3} + F(x) = G^2(x) \left(\frac{\partial \psi}{\partial y} \frac{\partial^2 \psi}{\partial x \partial y} - \frac{\partial \psi}{\partial x} \frac{\partial^2 \psi}{\partial y^2} \right), \tag{29}$$

$$\frac{\partial^2 \bar{\phi}}{\partial y^2} = P_r G^2(x) \left(\frac{\partial \psi}{\partial y} \frac{\partial \bar{\phi}}{\partial x} - \frac{\partial \psi}{\partial x} \frac{\partial \bar{\phi}}{\partial y} \right)$$
(30)

subject to boundary conditions

$$\psi = 0, \quad \frac{\partial \psi}{\partial y} = 0, \quad \bar{\phi} = 0 \quad \text{on } y = 0, \quad 0 \le x \le x_s,$$
 (31)

$$\psi = \gamma$$
, $\frac{\partial^2 \psi}{\partial y^2} = 0$, $\frac{\partial \phi}{\partial y} = 0$ at $y = \bar{h}(x)$, $0 \le x \le x_s$, (32)

$$\bar{h} = \gamma, \quad \psi = y, \quad \bar{\phi} = 1 \quad \text{on } x = 0, \quad 0 < y \leq \gamma.$$
 (33)

Accordingly the variables are transformed from (x, y) to (ξ, η) by using

$$x = \xi^{\alpha}, \qquad y = \frac{\xi \eta \overline{h}}{\xi + 1 - \eta}.$$

The dependent variables are transformed as

$$\begin{split} \psi &= \frac{\xi}{\xi + 1 - \eta} f, \quad \bar{u} = \frac{1}{(1 + \xi)^{\beta}} u, \quad \bar{v} = \frac{\xi + 1 - \eta}{\xi (1 + \xi)^{2\beta}} v, \quad \bar{\phi} = \phi, \\ \bar{w} &= \frac{\xi + 1 - \eta}{\xi (1 + \xi)^{\beta}} w, \quad \bar{h} = (1 + \xi)^{\beta} h. \end{split}$$

Table 1

Film thickness, free surface velocity and temperature for the cylinder with $F_r = 1$, $\gamma = 1$ and $P_r = 2$

x	film thickness	free surface	free surface
	$\bar{h}(x)$	velocity	temperature
		$\bar{u}(x,\bar{h}(x))$	$\bar{\phi}(x,\bar{h}(x))$
0.000	1.000	1.000	1.000
1.115×10^{-2}	1.182	1.000	1.000
3.361×10^{-2}	1.314	1.000	1.000
6.265×10^{-2}	1.426	0.998	1.000
0.105	1.541	0.981	0.999
0.211	1.717	0.914	0.983
0.301	1.796	0.868	0.957
0.501	1.836	0.823	0.885
0.714	1.767	0.840	0.801
1.054	1.611	0.917	0.665
1.303	1.527	0.972	0.573
1.588	1.474	1.013	0.478
2.029	1.472	1.021	0.359
2.545	1.608	0.946	0.261
2.985	1.953	0.800	0.206
π	2.218	0.722	0.191

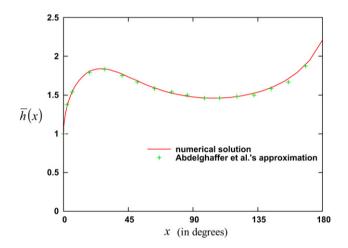


Fig. 2. Film thickness for the numerical solution and Abdelghaffer et al.'s approximation at $F_r = 1$ and $\gamma = 1$.

A full account of the numerical method and the details of implementation are beyond the scope of this paper and will be reported separately [12]. For the cylinder in this paper, the relevant physical parameters should be chosen as

$$F(x) = \frac{1}{F_r} \sin x, \quad G(x) = 1, \quad x_s = \pi, \quad \gamma = \frac{R_e^{1/2} H_0}{a},$$

 $\alpha = 2, \quad \beta = 0.$

The solution scheme was successfully tested against previously reported results [13–20].

5. Results

A typical run has a coarse grid of dimensions 60×48 in the (ξ, η) domain with each cell being divided into 1, 2, 3 and 4 subcells respectively. Because of the coordinate singularity at $\xi = 0$, $\eta = 1$, a non-uniform grid is employed and given by $\xi = \overline{\xi}^{1.5}$, $\eta = 1 - (1 - \overline{\eta})^{1.5}$ where $\overline{\xi}$ and $\overline{\eta}$ are uniform. When $\Delta \overline{\xi} \equiv \frac{1}{59}\pi^{\frac{1}{3}}$ and $\Delta \overline{\eta} \equiv \frac{1}{47}$, this gives $\Delta \xi \sim 0.004$ and $\Delta \eta \sim 0.003$ near the singularity, which is sufficiently small to give good accuracy. From the convergence of the extrapolation process previously described the absolute error is 2×10^{-7} . A typical set of numerical data is presented in Table 1.

In Fig. 2, the film thickness distribution around the cylinder is plotted from the numerical solution for $F_r = 1$ and $\gamma = 1$ and

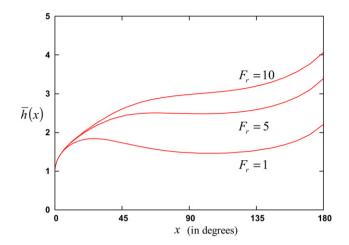


Fig. 3. Film thickness for various Froude numbers at $\gamma = 1$.

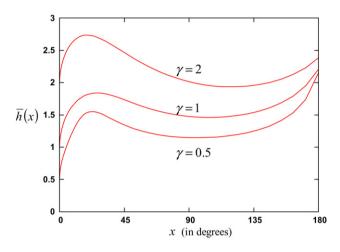


Fig. 4. Film thickness for various values of the parameter γ at $F_r = 1$.

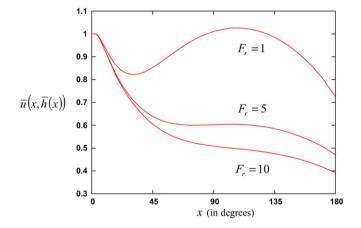


Fig. 5. Free surface velocity for various Froude numbers at $\gamma = 1$.

individual points calculated by Abdelghaffer et al. [8] using the approximate theory are included on the graph. The agreement is seen to be remarkably good. Figs. 3–9 depict the flavour of the numerical results. Figs. 3–4, 5–6 and 7–9 show film thickness, free surface velocity and free surface temperature respectively for various cases.

For the cylindrical case, the velocity of the flow is controlled by two opposing forces, viscosity trying to slow it down and gravity trying to speed it up. The gravitational component of force affect-

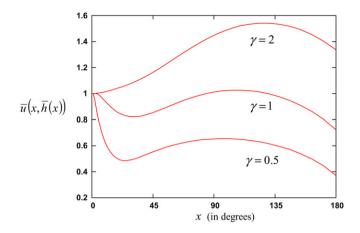


Fig. 6. Free surface velocity for various values of the parameter γ at $F_r = 1$.

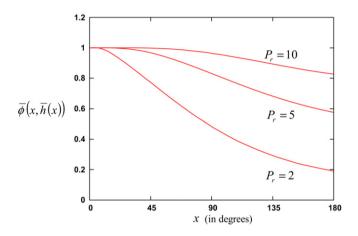


Fig. 7. Free surface temperature for various Prandtl numbers at $F_r = 1$ and $\gamma = 1$.

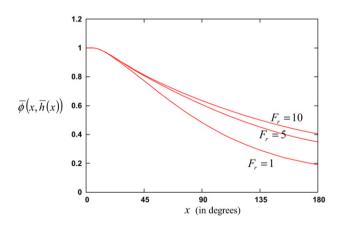


Fig. 8. Free surface temperature for various Froude numbers at $\gamma = 1$ and $P_r = 2$.

ing the flow is greatest near $x = \frac{\pi}{2}$ and least near x = 0 and $x = \pi$. Figs. 5–6 show that free surface velocity initially has a slight decrease, followed by a sharp increase as gravity starts to dominate, and finally a gradual decrease is observed as the bottom of the cylinder is approached. This corresponds to the up-down-up situation for the film thickness in Figs. 2–4, and the decline at the different rates for the free surface temperature in Figs. 7–9. As F_r decreases, the effect of gravity increases and hence the thin film thickness, high velocity and low temperature appear corresponding to the small F_r values. As γ decreases, the amount of fluid in the impinging sheet decreases and the ensuing film becomes thinner.

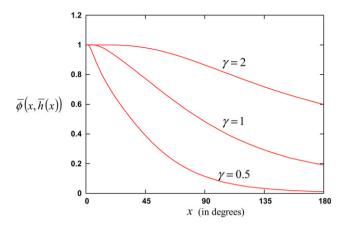


Fig. 9. Free surface temperature for various values of the parameter γ at $F_r = 1$ and $P_r = 2$.

The effect of initial viscosity increases and hence the initial velocity and temperature decreases become sharper as γ decreases. When γ is greater than a certain value, e.g., $\gamma = 2$ for $F_r = 1$, Fig. 6 shows no decrease for the initial velocity. As P_r increases, the temperature decrease becomes more gradual.

6. Concluding remarks

A detailed examination of a draining sheet flow over a horizontal cylinder has been performed. Comprehensive numerical solutions for establishing the flow and heat transfer characteristics of a cold, two-dimensional draining sheet over a hot, horizontal cylinder have been presented. The gross features of such flows have been illustrated over a range of representative parameter values. These indicate the underlying features of the developing film thickness, velocity and temperature distributions. In a practical setting, appropriate parameter values may be evaluated and the design characteristics readily identified from the numerical solutions. In practice, it is not obvious that uniform wetting of the cylinder would occur. Instabilities may distort or even disrupt such a uniform distribution. Nevertheless for a given overall flow rate, the model may represent a valuable first approximation to the aggregate properties of the flow.

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